FTR - Flight Test Report / Tandem Trimmer: geschlossen/closed

Manufacturer	SOL	Type testing No.	EAPR-GS-0655/17
	SOL SPORTS Rua Walter Marquardt, 1180 Jaraguá do Sul/SC - Brasil	serial number	19102
Model	Kuat 2	Location	Rofan, Achensee
Comment		Location	Rofan



Rev. 2.3 - 26.11.2014 EAPR GmbH - Marktstr. 11 D-87730 Bad Grönenbach - Germany

Date of testing	18.05.2017	Minimum take off weight 140 kg		Maximum take off weight 220 kg			
Testpilot		Anselm Rauh			Pascal Purin		
Harness		EAPR schwer			EAPR Tandem		
Pilot's take off weig	nt	138	kg	N. A. Yang	170/22	kg 🎒	7///





Test-criteria	st-criteria		Evaluation	Maximum take off weight	Evaluation	
1. Inflation / take-off - 4.4.1						
Rising behavior		Smooth, easy and constant rising,	А	Smooth, easy and constant rising,	А	
Special take off technique required		no pilot correction required No	Α	no pilot correction required No	A	
2. Landing - 4.4.2		140	Α	NO	Α	
Special landing technique required		I No	A	No	A	
3. Speeds in straight flight - 4.4.3		140	Α	140	Α	
Trim speed more than 30km/h		Yes	A	Yes	A	
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Speed range using the controls larger than 10km	√h	Yes	Α	Yes	A B	
Minimum speed		Less than 25 km/h	Α	25 km/h to 30 km/h		
4. Control movement - 4.4.4						
Max. weight in flight up to 80kg		Increasing > 65cm	А	Increasing > 65cm	А	
7. Roll stability and damping - 4.4.7						
Oscillations		Reducing	А	Reducing	Α	
8. Stability in gentle spirals - 4.4.8						
Tendency to return to straight flight		Spontaneous exit	А	Spontaneous exit	А	
9. Behaviour exiting a fully developed spiral	dive - 4.4	.9				
Initial response of glider (first 180°)		Immediate reduction of rate in turn	Α	Immediate reduction of rate in turn	A	
Initial response of glider (first 180°) Tendency to return to straight flight		Spontaneous exit	Α	Spontaneous exit	А	
Turn angle to recover normal flight		Less than 720°, spontaneous recovery	Α	720° to 1080°, spontaneous recovery	В	
10. Symmetric front collapse - 4.4.10						
Folding lines used		No		No		
Entry	8	Rocking back less than 45°	Α	Rocking back less than 45°	Α	
Recovery	90 ~ 30%	Spontaneous in less than 3 sec	А	Spontaneous in less than 3 sec	А	
Dive forward angle on exit	peeds	0° - 30° Keeping course	Α	0° - 30° Keeping course	Α	
Cascade occurs	frin	No	Α	No	Α	
Entry	> 50%	Rocking back less than 45°	Α	Rocking back less than 45°	Α	
Recovery	skeed > 5	Spontaneous in less than 3 sec	Α	Spontaneous in less than 3 sec	Α	
Dive forward angle on exit	8	0° - 30° Keeping course	Α	30° - 60° Keeping course	В	
Cascade occurs	ri i	No	А	No	Α	
11. Exiting deep stall (parachutal stall) - 4.4.	11					
Deep stall achieved				Yes		
Recovery		Spontaneous in less than 3 sec	А	Spontaneous in less than 3 sec	А	
Dive forward angle on exit		0° - 30°	Α	0° - 30°	Α	
Change of course		Changing course less than 45°	A	Changing course less than 45°	A	
Cascade occurs		No	Α	No	Α	
12. High angle of attack recovery - 4.4.12						
Recovery		Spontaneous in less than 3 sec	Α	Spontaneous in less than 3 sec	Α	
Cascade occurs		No A No		No	А	
13. Recovery from a developed full stall - 4.	1.13					
Dive forward angle on exit		0° - 30°	A	0° - 30°	A	
Collapse		No collapse	A	No collapse	A	
Cascade occurs (other than collapse)		No Less than 45°	A	No Less than 45°	A	
Rocking backward Line tension						

Folding lines used		No				No			1
•	olding lines used								
Change of course until re-inflation	Se	< 90°	Dive or roll angle	15° - 45°	Α	< 90°	Dive or roll angle	15° - 45°	Α
Re-inflation behavior	trim speed, max 50% collapse	Spontaneous re-inflation			Α	Spontaneous re-inflation			Α
Total change of course	u st	Less than 360)°		Α	Less than 360° No No			А
Collapse on the opposite side occurs	ž ţi.	No			Α				А
Twist occurs	ma	No			Α				А
Cascade occurs		No			A	No			A
Change of course until re-inflation	Ф	< 90°	Dive or roll angle	15° - 45°	Α	90° - 180°	Dive or roll angle	15° - 45°	В
Re-inflation behavior	trim speed, max 75% collapse	Spontaneous	re-inflation		A	Spontaneous re-inflation			А
Total change of course	sbe	Less than 360	10		Α	Less than 360°			А
	-ii 75		,						
Collapse on the opposite side occurs	ay t	No			A	No			A
Twist occurs		No			Α	No			Α
Cascade occurs		No			Α	No			Α
15. Directional control with a maintained asy	mmetric co	ollapse - 4.4.15							
Able to keep course straight		Yes			Α	Yes			А
180° turn away from the collapsed side possible	in 10 sec	Yes			Α	Yes			А
Amount of control range between turn and stall or spin		More than 509	nan 50% of the symmetric control travel A More than 50% of the symmetric control			control travel	Α		
16. Trim speed spin tendency - 4.4.16									
Spin occurs		No			Α	No			Α
17. Low speed spin tendency - 4.4.17									
Spin occurs		No			Α	l No			А
18. Recovery from a developed spin - 4.4.18	3	140				140			
Spin rotation angle after release		Stops spinning	g in less than 90°		А	Stops spinning	Stops spinning in less than 90°		
Cascade occurs	<u> </u>				A	No.			A
19. B-line-stall - 4.4.19		No				140			
Change of course before release		Changing cou	course less than 45° A Changing course less than 45°				Α		
Behaviour before release		Remains stable with straight span		Α	Remains stable with straight span			Α	
Recovery		Spontaneous in less than 3 sec			Α	Spontaneous in less than 3 sec			Α
Dive forward angle on exit		0° - 30°			Α	0° - 30°			А
Cascade occurs		No			Α	No			Α
20. Big ears - 4.4.20									
Entry procedure	Entry procedure		Special device required		Α	Special device required			Α
Behaviour during big ears		Stable flight			Α	Stable flight	Stable flight		
Recovery		Spontaneous	in 3 to 5 sec		В	Spontaneous in	Spontaneous in less than 3 sec		
Dive forward angle on exit		0° - 30°			Α	0° bis 30°			
23. Alternative means of directional control	- 4.4.22								
180° turn achievable in 20 sec		Yes			Α	Yes			Α
Stall or spin occurs		No			Α	No			А
23. Any other flight procedure and/or config	uration des	cribed in the us	er's manual - 4.4.2	23					
Procedure works as descibed					NA				N/
Procedure suitable for novice pilots		1			NA				N/
					NA				N/
Cascade occurs									
		<u> </u>			INA				

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